

1 **The Effect of Plant Weight on Estimations of Stalk Flexural Stiffness and Stalk**  
2 **Lodging Resistance**

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16 **Abstract**

17 **Background:** Stalk lodging (breaking of agricultural plant stalks prior to harvest) is a multi-  
18 billion dollar a year problem. Stalk lodging occurs when bending moments induced by a  
19 combination of external loading (e.g. wind) and self-loading (e.g. the plant's own weight) exceed  
20 the bending strength of plant stems. Previous studies have investigated external loading and self-  
21 loading of plants as separate and independent phenomena. However, these two types of loading  
22 are highly interconnected and mutually dependent. The purpose of this paper is twofold: (1) to  
23 investigate the combined effect of external loads and plant weight on the flexural response of  
24 plant stems, and (2) to provide a generalized framework for accounting for self-weight during  
25 mechanical phenotyping experiments used to predict stalk lodging resistance.

26 **Results:** A method of properly accounting for the interconnected relationship between self-loading  
27 and external loading of plants stems is presented. The interconnected set of equations are used to  
28 produce user-friendly applications by presenting (1) simplified self-loading correction factors for  
29 a number of common external loading configurations of plants, and (2) a generalized Microsoft  
30 Excel framework that calculates the influence of self-loading on crop stems. The effect of self-  
31 loading on the bending strength measurements of wheat is examined in detail. A survey of several  
32 other plants is conducted and the influence of self-loading on their structural response is also  
33 presented.

34 **Conclusions:** The self-loading of plants plays a potentially critical role in determining the  
35 structural response of plant stems. Equations and tools provided herein enable researchers to  
36 account for the plant's weight during mechanical phenotyping experiments used to determine the  
37 flexural rigidity and bending strength of plant stems. Results demonstrated that ignoring the self-  
38 loading of some plants can result in errors of 25% for flexural stiffness and 20% for bending  
39 strength.

40

41 **Keywords:** bending, biomechanics, computational, flexural, plant, lodging, stalk, stem, stiffness,  
42 strength, weight

## 43 **Background**

44 Yield losses due to stalk lodging (breakage of crop stems or stalks prior to harvest) are  
45 estimated to range from 5-20% annually [1,2]. Despite a growing body of literature surrounding  
46 the topic of stalk lodging in wheat, barley, oats, and maize [3–7], a detailed study on the  
47 interconnected relationship between external loading (e.g. wind) and self-loading (e.g. plant  
48 weight) on stalk bending strength and stalk flexural stiffness has not been reported. In particular,  
49 previous biomechanical models of stalk lodging have examined the influence of morphology,  
50 material, and weight on stem failure, while others have separately analyzed the effects of externally  
51 induced bending forces (e.g., wind) on stem failure [3,4,8–15]. However, the effects of self-weight  
52 and external loads on stem failure are inextricably connected. The bending moment induced from  
53 self-weight is a function of the distance between the plant’s base and its center of gravity. As  
54 external loads displace the center of gravity away from the base of the stem, the bending moment  
55 induced from self-weight increases. Although previous studies have independently looked at  
56 external loads [4,5,16] and self-weight loading [3], the authors are not aware of previous  
57 investigations into the interplay between these factors. A more complete understanding of the  
58 interconnected relationship between self-loading and external loading is necessary to improve the  
59 accuracy of mechanical phenotyping experiments used to determine stalk flexural stiffness and  
60 stalk bending strength (i.e., measurements of stalk lodging resistance).

61 Several phenotyping devices have been developed to assess the stalk lodging resistance of  
62 grain crops [24]. In general, these devices apply an external load to either a single plant or to a  
63 group of plants and measure the accompanying deflection of the plant stem(s). A biomechanical  
64 model is then used to determine the flexural stiffness or bending strength of the plant sample(s).  
65 Measurements of flexural stiffness and bending strength are highly correlated with natural

66 occurring stalk lodging and are therefore often used as surrogate measurements of stalk lodging  
67 resistance. Although biomechanical models of plants can vary widely, a typical biomechanical  
68 model of a grain plant undergoing stalk lodging will consider the stem as a cantilever beam in  
69 bending, as shown in Figure 1. The loads applied to this beam can vary from a single point load  
70 at the top of the stem, to a distributed load applied along the entire length of the stem as shown in  
71 [22]. Standard engineering beam equations are then used to calculate the mechanical properties  
72 of the stem (e.g., flexural stiffness and bending strength). For example, the device developed by  
73 Cook *et al.* [6,41] laterally pushes a maize stalk, measuring the force applied to the stem and its  
74 deflected angle. Then, leveraging the cantilever beam model equations, the flexural stiffness and  
75 the bending strength of the stem is calculated. However, these calculations ignore the effect of  
76 self-weight and thus underpredict the actual strength / stiffness of the plant.

77 To more accurately calculate stem bending strength and flexural stiffness from mechanical  
78 phenotyping data it is necessary to calculate the total applied bending moment ( $M_{TOTAL}$ ) to the plant.  
79 The total applied bending moment is equal to the summation of externally applied bending loads  
80 ( $M_{ext}$ ) and bending loads induced from self-weight ( $M_{int}$ ). It should be noted that externally applied  
81 bending loads can arise from any externally applied force (e.g., the wind, mechanical testing  
82 equipment, an adjacent plant pushing on the plant of interest or loads applied from a phenotyping  
83 device). However, bending loads induced from self-weight can only arise from the weight of the  
84 plant itself. The equations developed in this study can be used to account for any of the external  
85 loads mentioned above. Let us first consider the case of a mechanical phenotyping device used to  
86 assess stalk lodging resistance. As mentioned previously a typical phenotyping device for  
87 assessing stalk lodging resistance of grain crops applies an external bending load ( $M_{ext}$ ) to a plant  
88 stem and measures its deflection ( $\delta$ ) [24,42,43]. The externally applied moment ( $M_{ext}$ ) and plant

89 displacement ( $\delta$ ) data are then analyzed to determine the flexural stiffness (EI) of the stem or the  
90 bending strength of the stem using Castigliano's energy method, where the displacement of the  
91 beam is equal to the partial derivative of the internal potential energy of the system with respect to  
92 the applied load. For example see [17]:

$$EI = \frac{\int M_{ext} \frac{dM_{ext}}{dQ} dx}{2\delta} \quad (1)$$

93  
94 where Q is the dummy load used for the load-moment derivative [17]. The effect of self-weight  
95 on plant deflection is ignored or assumed to be negligible in these analyses. In other words this  
96 type of analysis assumes bending strength =  $M_{TOTAL} = M_{ext}$ , whereas in reality bending strength =  
97  $M_{TOTAL} = M_{ext} + M_{int}$ . In some instances researchers have removed leaf blades, grain heads, and all  
98 plant biomass above the grain prior to mechanical testing to limit the effect of self-weighting (i.e.  
99 to minimize  $M_{int}$ ). However, it is unclear if such measures are adequate. To more accurately  
100 characterize flexural stiffness and stalk bending strength the contribution of self-weight to plant  
101 deflection and to the induced bending moment  $M_{TOTAL}$  needs to be calculated.

102 The purpose of this paper is therefore twofold: (1) to investigate the combined effect of  
103 external loads and plant weight on the bending strength and flexural stiffness of plant stems, and  
104 (2) to provide a generalized framework for accounting for self-weight during mechanical  
105 phenotyping experiments used to predict stalk lodging resistance.

106

## 107 **Methods**

108

### *Box 1: Glossary of Terms*

| <b>Term</b>               | <b>Definition</b>  |
|---------------------------|--|
| $\delta$                  | Horizontal deflection (mm)   |
| EI                        | Flexural stiffness ( $\text{mm}^2$ )                                       |
| F                         | Externally applied force (N)   |
| $f_M$                     | Geometric coefficient for applied moments                                  |
| $f_F$                     | Geometric coefficient for applied forces                                   |
| h                         | Height (mm)  |
| L                         | Location where loading is applied (mm)                                     |
| M                         | Externally applied moment (Nmm)  |
| $M_{\text{ext}}$          | Total induced moment from externally applied forces and moments (Nmm)      |
| $M_{\text{int}}$          | Total induced moment from self-loading (Nmm)                               |
| $M_{\text{TOTAL}}$        | Total applied moment (sum of $M_{\text{ext}}$ and $M_{\text{int}}$ ) (Nmm) |
| Q                         | The Castigliano's Method dummy load (N)                                    |
| S                         | Section modulus ( $\text{mm}^3$ )  |
| w                         | Weight (N)   |
| W                         | Weight-induced moment (Nmm)  |
| $\sigma_{\text{bending}}$ | Bending stress ( $\text{N}/\text{mm}^2$ )                                  |
| Z                         | Vertical position where deflection is being calculated (mm)                |

109

### 110 *Derivation of Closed Form Solution*

111 To determine the contribution of self-weight to the bending strength and flexural stiffness  
112 of plant stems, we must first derive a closed form solution for the internal bending moment of the  
113 stem ( $M_{\text{TOTAL}}$ ). Figure 1 depicts the free body diagram of a plant stem with an arbitrary loading  
114 applied at two locations. This stem depicts two weights (w) (e.g. stem weight, grain weight), as  
115 well as two externally applied loads (F) and two externally applied moments (M). Note that as  
116 mentioned before the externally applied loads and moments can be arise from any externally  
117 applied force. Commons sources of externally applied forces include phenotyping devices, wind,  
118 adjacent plants and mechanical testing equipment.

119

120 As the stem deflects, the moments induced from self-weights will increase as a function of  
121 the deflection of the stem. For the weight ( $w$ ) at each location, we can calculate the induced  
122 moment from self-weight ( $W$ ) as the product of the weight and the weight's offset (i.e., the  
123 deflection of the stem at the location of the weight ( $\delta$ )). Thus for the two locations shown in Figure  
124 1, we have:

$$125 \quad W_1 = \delta_1 w_1 \quad (2)$$

$$126 \quad W_2 = \delta_2 w_2 \quad (3)$$

127 It should be noted that Equations 2 and 3 assume that the maximum moment induced by  
128 self-loading is applied to the entire length of the stem. For more detail, see the Limitations  
129 section.

130 However, the offsets ( $\delta_1$  and  $\delta_2$ ) in equations 2 and 3 are not known and are a function of  
131 the externally applied moments and forces. Using engineering theory for beam deflection and  
132 the theory of superposition of loading [17], we can calculate the deflection of the stem at height  
133  $h_1$  (i.e., location 1) as a function of the applied forces, applied moments, and weight-induced  
134 moments. Equation 4 shows this calculation, where the first row of equation 4 concerns loads,  
135 moments and weights at location 1 (i.e., at height  $h_1$ ) and the second row of equation 4 concerns  
136 forces, moments and weights at location 2 (i.e., at height  $h_2$ ).

|              | Applied Forces                   | Applied Moments                     | Weight-Induced Moments              |         |
|--------------|----------------------------------|-------------------------------------|-------------------------------------|---------|
| $\delta_1 =$ | $F_1 \cdot \frac{h_1^3}{3EI}$    | $+ M_1 \cdot \frac{h_1^2}{2EI}$     | $+ W_1 \cdot \frac{h_1^2}{2EI}$     | Loc. #1 |
| 137          | $F_2 \frac{3h_1h_2 - h_2^2}{EI}$ | $+ M_2 \frac{2h_1h_2 - h_2^2}{2EI}$ | $+ W_2 \frac{2h_1h_2 - h_2^2}{2EI}$ | Loc. #2 |

(4)

138 Similarly, we can write the deflection of the stem at  $h_2$  as:

|              | Applied Forces                            | Applied Moments                 | Weight-Induced Moments          |         |
|--------------|---|---------------------------------|---------------------------------|---------|
| $\delta_2 =$ | $F_1 \cdot \frac{h_1^2(3h_1 - h_2)}{6EI}$ | $+ M_1 \cdot \frac{h_2^2}{2EI}$ | $+ W_1 \cdot \frac{h_1^2}{2EI}$ | Loc. #1 |
| 139          | $+ F_2 \cdot \frac{h_2^3}{3EI}$           | $+ M_2 \cdot \frac{h_2^2}{2EI}$ | $+ W_2 \cdot \frac{h_2^2}{2EI}$ | Loc. #2 |

(5)

140

141 Thus we have four linearly independent equations (Equations 2 through 5) allowing us to  
 142 solve for four unknown values ( $W_1, W_2, \delta_1, \delta_2$ ). It should be emphasized that for all equations in  
 143 this manuscript (including equations 4 and 5) locations are numbered from the top of plant down  
 144 (i.e., location 1 is above location 2 which is above location 3...)

145 Equations 2 through 5 can be generalized to account for any number of locations (n) along  
 146 the length of the stalk. First, for any loading location L, at a height  $h_L$  along the stalk, deflected by  
 147  $\delta_L$ , Equations 2 and 3 can be generalized as:

$$148 \quad W_L = \delta_L W_L \tag{6}$$



149 Next, Equations 4 and 5 can be generalized by noting that each force, moment or weight  
 150 (F, M, or W, shown in bold in Equations 4 and 5) is multiplied by a geometric coefficient. The  
 151 geometric coefficient for each term is a function of the height where the deflection is measured  
 152 and the height at which the loading is applied. This geometric coefficient can be denoted as either  
 153  $f_F$  (for forces) or  $f_M$  (for applied moments or weight-induced moments). As such, for any vertical  
 154 location Z at a height of  $h_p$ , the deflection  $\delta_p$  is calculated by summing the product of each load,  
 155 moment or weight (F, M, or W) and its corresponding geometric coefficient ( $f_F$  or  $f_M$ ) at every  
 156 loading location (from L=1 to L=n). Note that this geometric coefficient assumes a constant  
 157 flexural stiffness (EI), as discussed in the Limitations section. Thus the generalized form of  
 158 Equations 4 and 5 can be written as:

$$\begin{array}{ccccccc}
 & \text{Applied Forces} & & \text{Applied Moments} & & \text{Weight-Induced Moments} & \\
 \delta_p = & \mathbf{F}_1 \cdot f_F(Z, 1) & + & \mathbf{M}_1 \cdot f_M(Z, 1) & + & \mathbf{W}_1 \cdot f_M(Z, 1) & \text{Loc. \#1} \\
 & + & \mathbf{F}_2 \cdot f_F(Z, 2) & + & \mathbf{M}_2 \cdot f_M(Z, 2) & + & \mathbf{W}_2 \cdot f_M(Z, 2) & \text{Loc. \#2} \\
 & & \vdots & & \vdots & & \vdots & \\
 & + & \mathbf{F}_L \cdot f_F(Z, n) & + & \mathbf{M}_L \cdot f_M(Z, n) & + & \mathbf{W}_L \cdot f_M(Z, n) & \text{Loc. \#L}
 \end{array} \tag{7}$$

159  
 160 Where “location 1” is the most apical location of interest and “location L” is the most basal location  
 161 of interest. Equation 7 can now be consolidated into a fully generalized form of:

$$\begin{array}{ccc}
\text{Applied Forces} & \text{Applied Moments} & \text{Weight-Induced Moments} \\
\delta_p = \sum_{L=1}^n F_L f_F(Z, L) & + \sum_{L=1}^n M_L f_M(Z, L) & + \sum_{L=1}^n W_L f_M(Z, L)
\end{array}
\tag{8}$$

162

163 Where the geometric coefficients for the forces and moments are defined as [18]:

$$f_F(P, L) = \begin{cases} 3h_L h_p^2 - \frac{(h_L - h_p)^3}{6EI}, & h_p \geq h_L \\ \frac{h_L^2(3h_L - h_p)}{6EI}, & h_p < h_L \end{cases}
\tag{9}$$

164

$$f_M(P, L) = \begin{cases} \frac{h_L(2h_p - h_L)}{2EI}, & h_p \geq h_L \\ \frac{h_p^2}{2EI}, & h_p < h_L \end{cases}
\tag{10}$$

165

166 Equations 6 through 9 can also be put into a generalized matrix form. From Equations 6  
167 and 8 we see that for any number of weights at any number of locations (n), we will have 2n  
168 unknown values ( $\delta_1, \delta_2, \dots, \delta_n, W_1, W_2, \dots, W_n$ ), and 2n linearly independent equations. By  
169 rearranging these equations and converting them to matrix notation we can write:

$$\begin{bmatrix}
1 & 0 & \dots & 0 & -f_M(1,1) & -f_M(1,2) & \dots & -f_M(1,n) \\
0 & 1 & \dots & 0 & -f_M(2,1) & -f_M(2,2) & \dots & -f_M(2,n) \\
\vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \ddots & \vdots \\
0 & 0 & \dots & 1 & -f_M(n,1) & -f_M(n,2) & \dots & -f_M(n,n) \\
-w_1 & 0 & \dots & 0 & 1 & 0 & \dots & 0 \\
0 & -w_2 & \dots & 0 & 0 & 1 & \dots & 0 \\
\vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \ddots & \vdots \\
0 & 0 & \dots & -w_n & 0 & 0 & \dots & 1
\end{bmatrix}
\begin{bmatrix}
\delta_1 \\
\delta_2 \\
\vdots \\
\delta_n \\
W_1 \\
W_2 \\
\vdots \\
W_n
\end{bmatrix}
=
\begin{bmatrix}
\sum_{L=1}^n F_{Lf_F}(1,L) + \sum_{L=1}^n M_{Lf_M}(1,L) \\
\sum_{L=1}^n F_{Lf_F}(2,L) + \sum_{L=1}^n M_{Lf_M}(2,L) \\
\vdots \\
\sum_{L=1}^n F_{Lf_F}(n,L) + \sum_{L=1}^n M_{Lf_M}(n,L) \\
0 \\
0 \\
\vdots \\
0
\end{bmatrix}
\tag{11}$$

170  
171 Where the first matrix in the equation is a square matrix of size  $2n \times 2n$ , and the second and third  
172 matrices in the equation are column matrices of size  $2n \times 1$ . Within the square matrix, the top  
173 left and bottom right  $n \times n$  submatrices (shown in green text) are identity matrices, the bottom  
174 left  $n \times n$  submatrix (shown in blue text) is a diagonal matrix of the negative weights (-w), and  
175 the top right  $n \times n$  submatrix (shown in orange text) is the negative geometric coefficients of the  
176 weight-induced moments, as calculated by Equation 10. We can then solve this matrix equation  
177 by taking the inverse of the multi-colored matrix and multiplying by the right-most vector to  
178 calculate the deflections and weight-induced moments:

$$\begin{bmatrix}
\delta_1 \\
\delta_2 \\
\vdots \\
\delta_n \\
W_1 \\
W_2 \\
\vdots \\
W_n
\end{bmatrix}
=
\begin{bmatrix}
1 & 0 & \dots & 0 & -f_M(1,1) & -f_M(1,2) & \dots & -f_M(1,n) \\
0 & 1 & \dots & 0 & -f_M(2,1) & -f_M(2,2) & \dots & -f_M(2,n) \\
\vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \ddots & \vdots \\
0 & 0 & \dots & 1 & -f_M(n,1) & -f_M(n,2) & \dots & -f_M(n,n) \\
-w_1 & 0 & \dots & 0 & 1 & 0 & \dots & 0 \\
0 & -w_2 & \dots & 0 & 0 & 1 & \dots & 0 \\
\vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \ddots & \vdots \\
0 & 0 & \dots & -w_n & 0 & 0 & \dots & 1
\end{bmatrix}^{-1}
\begin{bmatrix}
\sum_{L=1}^n F_{Lf_F}(1,L) + \sum_{L=1}^n M_{Lf_M}(1,L) \\
\sum_{L=1}^n F_{Lf_F}(2,L) + \sum_{L=1}^n M_{Lf_M}(2,L) \\
\vdots \\
\sum_{L=1}^n F_{Lf_F}(n,L) + \sum_{L=1}^n M_{Lf_M}(n,L) \\
0 \\
0 \\
\vdots \\
0
\end{bmatrix}
\tag{12}$$

179

180 We can now look at the total moment ( $M_{TOTAL}$ ) of any cross-section along the length of the stem.  
 181 In particular,  $M_{TOTAL}$  can be written as a function of  $h_p$  and  $h_L$ , by considering all of the loads that  
 182 are applied to the stem above the cross-section of interest (i.e, for  $h_L \geq h_p$ ),

$$M_{TOTAL}(h_p) = \sum_{L=1}^{n [h_L \geq h_p]} F_L(h_L - h_p) + \sum_{L=1}^{n [h_L \geq h_p]} M_L + \sum_{L=1}^{n [h_L \geq h_p]} W_L$$

183 (13)

184 Finally, we can write the bending stress of the stem in terms of the internal moment and the  
 185 section modulus of the cross-section ( $S(h_z)$ ):

$$\sigma_{bending}(h_p) = \frac{M_{TOTAL}(h_p)}{S(h_p)}$$

186 (14)

187 Note that “section modulus” is an engineering term used to quantify the cross-sectional distribution  
 188 of mass about its centroid and is used in making flexural stiffness and bending strength calculations  
 189 [17].

### 190 *Finite Element Modeling and Data Triangulation*

191 As a form of data triangulation [18] to confirm no mistakes were made in the derivation of  
 192 the closed form solution presented above, a series of 768 non-linear finite element models of plant  
 193 stems were developed. Finite element models are a numerical technique used by engineers to  
 194 quantify the detailed mechanical response of complex structures and materials [44]. It should be  
 195 noted that nonlinear finite element models (i.e. “large displacement” simulations) are valid for  
 196 both small and large displacements. In this simulations the stems were modeled as 2-noded linear  
 197 beam elements in a 2-dimensional analysis [19,20]. The bottom node of the stem model was fixed

198 in all degrees of freedom ( $U_1 = U_2 = U_3 = 0$ ). A mesh convergence study was performed to  
199 ensure adequate mesh density. Analyses were run non-linearly, recalculating the system stiffness  
200 matrix at each solution increment. In other words, the models were fully capable of accounting  
201 for nonlinear effects due to large deformations. The models were developed in Abaqus/CAE 2019  
202 [19,20] and analyzed in Abaqus/Standard 2019 using a direct, full Newton solver [19,20]. Model  
203 development and post-processing were automated through a series of custom Python  
204 scripts. Python scripts can be obtained upon reasonable request to the authors. Stems were  
205 modeled with a weight at height  $h_1$ , applied force at height  $h_2$ , and applied moment at height  $h_3$ . It  
206 should be noted that because 2-noded beam elements were used, the model was partitioned at  $h_3$   
207 so that moments could be directly applied to nodes. The beam was modeled with the radius values  
208 such that the resulting moment of inertia were as presented in Table 1 using the equation  $I =$   
209  $\frac{\pi}{4} r^4$  [17]. A full parametric sweep of all relevant input parameters (i.e. factors) to the model was  
210 conducted. In particular, a factorial design of experiments was utilized with 8 factors. The factors  
211 were the elastic moduli of the beam ( $E$ ), the moment of inertia ( $I$ ) of the beam, the heights of the  
212 applied moments, forces and weights ( $h_1$ ,  $h_2$  and  $h_3$ ), the magnitude of the applied moment ( $M$ ),  
213 the magnitude of self-weight ( $W$ ), and the magnitude of the applied force ( $F$ ). The moduli, moment  
214 of inertia, heights, weights, and moments were evaluated at two different levels. The force was  
215 evaluated at 6 levels. Thus a total of 768 unique models were constructed covering every  
216 combination of factors and levels (i.e.,  $2E$ 's  $\times$   $2I$ 's  $\times$   $2h_1$ 's  $\times$   $2h_2$ 's  $\times$   $2h_3$ 's  $\times$   $2M$ 's  $\times$   $2W$ 's  $\times$   $6F$ 's  
217  $=$  768 models). Table 1 presents each of these factors and the levels of each factor. The level of  
218 each factor (i.e., the value of input parameters to the model) were based on previous studies of  
219 plant stem material properties [21, 22]. The total moment ( $M_{total}$ ) at the base of the beam and the

220 deflection at the top of the beam for each finite element model were then compared to the  
 221 deflection and total moment ( $M_{total}$ ) calculated using the closed form solution presented above.

222 **Table 1:** Each input parameter (i.e., factor) and value of each input parameter (i.e., level) for the  
 223 finite element analyses. The number of levels for each factor noted as (n) is presented in the  
 224 bottom row of the table. The force (F) had 6 levels (0, 2, 4, 6, 8, and 10 newtons). All other  
 225 factors had 2 levels (a maximum value and a minimum value). A total of 768 finite element  
 226 models were evaluated ( $2E's \times 2I's \times 2h_1's \times 2h_2's \times 2h_3's \times 2M's \times 2W's \times 6F's = 768$   
 227 models).

| Value   | E (N/mm <sup>2</sup> ) | I (mm <sup>4</sup> ) | h <sub>1</sub> (mm) | h <sub>2</sub> (mm) | h <sub>3</sub> (mm) | M (Nmm) | W (N) | F (N) |
|---------|------------------------|----------------------|---------------------|---------------------|---------------------|---------|-------|-------|
| Minimum | 1.00E+03               | 1.00E+04             | 800                 | 400                 | 100                 | 0       | 0     | 0     |
| Maximum | 1.00E+08               | 1.00E+05             | 1200                | 700                 | 300                 | 100     | 2     | 10    |
| n =     | 2                      | 2                    | 2                   | 2                   | 2                   | 2       | 2     | 6     |

228

### 229 *Analyzing the Effect of Self-Loading on Wheat*

230 To determine the effects of self loading on the bending strength and flexural stiffness of wheat  
 231 the closed form solution method presented above was applied to vertically-partitioned wheat  
 232 biomass data. Biomass data was collected from a commercially available wheat (*Triticum*  
 233 *aestivum*) variety during the 2018 growing season in Saskatoon, Saskatchewan. As depicted in  
 234 Figure 2 the biomass of wheat stems, leaves, and spikes were sampled and weighed every 10 cm  
 235 along the length of the plants. Planting density was approximately 1.3 million plants per hectare  
 236 with a 30.5 cm (12”) row spacing.

237 Biomass data were gathered from a 68 cm x 68 cm square of wheat in the center of a 122  
238 cm wide plot. The 68 cm x 68 cm square contained an average of 366 stems. A total of five  
239 samples of biomass data were taken periodically from July 27 to August 29, 2018. The same plot  
240 was used for all sampling dates with enough space left between samples to have undisturbed  
241 wheat in each subsequent sample. A square guide was placed over the middle rows of the plot to  
242 indicate the sampling area and any plants outside of the guide were then removed. Biomass was  
243 harvested in 10 cm layers measured from the ground with the highest layer collected first  
244 (topmost layer varied in size depending on total plant height). All plant matter from a single  
245 layer was harvested, weighed in the field to obtain wet-basis biomass, and bagged to be dried  
246 later. The samples were oven dried at 65°C for a minimum of 48 hours to obtain the dry-basis  
247 biomass.

248

## 249 **Results**

### 250 *Comparison of Finite Element and Closed Form Solutions*

251 As a form of data triangulation finite element models of plant stems were compared to the closed  
252 form solution presented in the methods section. In other words, the closed form solution was  
253 evaluated using the same inputs as each of the 768 finite models and the solutions from each set  
254 of input parameters were compared. The finite element models were found to be in good agreement  
255 with the closed form solutions (see Figure 3). In particular, the median error between the 768  
256 finite element models and the corresponding closed form solutions was found to be 0.126% for  
257 displacement at the top of the specimen, and 0.0003% for the total moment at the base of the  
258 specimen. These data imply that for the ranges evaluated, the closed form solution is providing  
259 accurate results and no mistakes were made during its derivation.

260 Standard engineering beam equations used to derive the closed form solution presented  
261 above contain several inherent assumptions. These assumptions gradually become less valid as  
262 deflections become very large. Therefore, to determine the maximum range of applicability for  
263 the closed form solutions one additional finite element model was created and subjected to  
264 extremely large deflections. In particular, the model was created with the following input  
265 parameters:  $E = 5.00E+07 \text{ Nmm}^2$ ,  $I = 5.50E+04$ ,  $EI = 2.8E12$ ,  $h_1 = 1000 \text{ mm}$ ,  $h_2 = 550 \text{ mm}$ ,  $h_3 =$   
266  $200 \text{ mm}$ ,  $M = 1000 \text{ Nmm @ } h_1$ ,  $W = 100 \text{ N @ } h_3$ ,  $F = \text{Ramped up to } 5.00E+07 \text{ @ } h_2$ . It should  
267 be noted that this loading scenario exceeds the realistic range of forces and deflections a plant stem  
268 would be subjected to before breaking. This extreme model was used to investigate the extent of  
269 validity of the closed form solution for very large deflections. Agreement between this finite  
270 element model and the closed form solutions is strong at small displacements (as expected). At  
271 large displacements (greater than  $\sim 45^\circ$  at the tip), geometric nonlinearities that are not captured by



272 the closed form engineering beam equations become more influential [4]. That is to say that the  
273 closed form solution is accurate so long as the linear closed form engineering beam equations upon  
274 which it is predicated are accurate. For more discussion on this topic, see the Limitations section.  
275 Figure 4 depicts the comparison between the extremely large deflection finite element model and  
276 the closed form solution. Figure 4 displays a maximum horizontal displacement equal to the height  
277 of the stem.

278

### 279 *A Computational Tool for Accounting for Weights*

280 To make the closed form solutions more amenable to researchers without a structural  
281 engineering background (i.e., plant scientists, agronomists, and other end-users), an Excel  
282 (Microsoft Corporation, 2019) spreadsheet was developed, and is included as Additional File  
283 1. The user simply inputs the flexural stiffness of the plant stem, the heights to each location of  
284 interest, the magnitude of externally applied forces and moments, and the weights at each  
285 location. Input values can be given for up to ten locations of interest along the length of the plant  
286 stem. The spreadsheet calculates the weight induced moments ( $M_{int}$ ) and displacements as well  
287 as the total induced moment ( $M_{total}$ ) at all locations. The spreadsheet makes the calculation both  
288 with and without self-loading considered. In addition, the error induced by ignoring the self-  
289 loading is calculated for the displacements and total induced moments. Additional instructions  
290 for using the spreadsheet can be found in the Additional File 2. Figure 5 shows an example of  
291 the spreadsheet in which 3 externally applied forces, 2 externally applied moments, and 3  
292 weights are considered. This tool can be used by researchers to determine the necessity of  
293 including self-loading in their studies.

294 For example, if this spreadsheet were used to determine the necessity of including self-  
295 weight in a mechanical phenotyping study (e.g., a study using the device as presented in [6]), the  
296 following would be performed: (1) A non-destructive, small displacement, flexural test as  
297 described in [6] would be performed, to determine the specimen's flexural stiffness; (2) a  
298 destructive, large displacement bending strength test as described in [6] would then be performed  
299 on the same specimen; (3) the specimen would then be weighed and the center-of-gravity would  
300 be determined; (4) the specimen weight, center-of-gravity, and flexural stiffness as well as the  
301 magnitude and location of the load applied to the plant by the phenotyping device from the  
302 destructive bending strength test would be input into the spreadsheet; (5) the spreadsheet would  
303 report out the amount of error in the total induced moment and displacement if the weight of the  
304 specimen was ignored. This procedure would then be repeated for several representative  
305 specimens. This data could then be used to inform the researchers if self-weight induced loadings  
306 are significant and need to be accounted for in phenotyping experiments or if the amount of error  
307 introduced by neglecting self-weight is negligible. If self-weight was determined to be  
308 significant then the spreadsheet could be used to properly account for self-weight of measured  
309 samples or alternatively the researchers could use the closed form solutions presented above.

### 310 ***Case Study: The Effect of Self-Loading on Wheat***

311 As a demonstrative case-study, the effect of self-loading on wheat was analyzed. The flexural  
312 stiffness of wheat was assumed to be  $0.027 \text{ Nm}^2$  for this analysis [23]. The wheat biomass data  
313 collected as part of the current study (see methods section) was used to apply weights to the stem  
314 at every 10 cm along the length of the plant. The biomass data are shown in Additional File 3.  
315 The effect of stem weight on the stems flexural response was then analyzed using the spreadsheet  
316 described above (see Additional Files 1 and 2). Figure 6 depicts the effect of self-loading on wheat

317 stems sampled in this study. The figure displays the contribution of self-loading to the total  
318 displacement at the top of the stem and to the total moment at the base of the stem as a  
319 percentage. The contribution of self-loading was found to increase over the growing season, and  
320 then decrease during senescence. The contribution of self-loading was found to be significantly  
321 more for wet stems than dry stems, which is to be expected. As shown in Figure 6 self-weight can  
322 account for up to 25% of the displacement of wheat stems and as much as 20% of the total moment.

323

## 324 **Discussion**

### 325 *The Influence of Self-Loading*

326 The relative contribution of self-weight to the total bending moment can be conceptualized based  
327 on two key characteristics: (1) the displacement of the stalk, and (2) the ratio of weight to flexural  
328 stiffness (EI). This is because the moment induced by the weight is directly related to its offset  
329 (i.e., displacement of the stalk), as shown in Equation 6. The offsets (i.e., stem displacement) are  
330 in turn dependent on the flexural stiffness of the stalk. This means that a stalk with high flexural  
331 stiffness that requires a larger applied force to increase its displacement will be relatively less  
332 influenced by self-weight. Conversely, a stalk with low flexural stiffness that requires a small  
333 amount of applied load to increase its displacement will be relatively more influenced by self-  
334 weight.

335 To aid researchers in determining if the influence of self-weight is a significant factor in a  
336 given experiment the following examples are presented. These examples pertain to common  
337 mechanical phenotyping tests used to quantify flexural stiffness [24]. In addition, we present

338 simple correction factors for the self-weight-induced moments that are typically ignored in such  
339 experiments.

340 The first example loading configuration, which represents a typical flexural stiffness test  
341 for maize [6,45], applies a point load at the top of the specimen, while the stalk's center of  
342 gravity is below the loading point. The second example loading configuration, which represents  
343 a typical flexural stiffness test for wheat [25, 26], applies a point load below the grain head but  
344 near the top of the specimen. Figure 7 displays the loading diagrams for these two test  
345 configurations.

346 During these mechanical phenotyping tests the applied load (F) and deflection of the stem  
347 at the point of loading ( $\delta_t$ ) is typically recorded. Ignoring the weight of the stalk, the flexural  
348 stiffness of the stem (EI) is then calculated from the test data by rearranging the following  
349 equation to solve for EI:

$$\delta_t = \frac{Fh_t^3}{3EI} \quad (15)$$

351 To account for the weight of the stalk when calculating the flexural stiffness of the stem, we  
352 must modify Equation 15 to include the stalk weight (w) as discussed in the methods section. For  
353 example:

354 *Configuration 1: Load at Top, Weight at Midspan*

356 First, solving Equation 11 for loading configuration 1 results in:

$$\begin{bmatrix} 1 & \frac{-h_2}{2EI} \\ -w & 1 \end{bmatrix} \begin{bmatrix} \delta_2 \\ W \end{bmatrix} = \begin{bmatrix} \frac{Fh_2^2(3h_1 - h_2)}{6EI} \\ 0 \end{bmatrix} \quad (16)$$

358 Where the two unknowns are the displacement at the weight ( $\delta_2$ ) and the weight-induced  
 359 moment (W). From this equation, the weight-induced moment can be calculated as:

$$360 \quad W = \frac{Fh_2^2 w(3h_1 - h_2)}{6EI - 3wh_2^2} \quad (17)$$

361 Finally, we can solve Equation 5 at the point of loading ( $\delta_1$ ) to find a relationship between  
 362 the test data and the flexural stiffness of the stem:

$$363 \quad \delta_1 = \frac{Fh_1}{3EI} + \frac{Fwh_2^2(3h_1 - h_2)(2h_1 - h_2)}{6EI(2EI - wh_2^2)} \quad (18)$$

364 Where Equation 15 is shown in black, and the correction factor for the weight-induced moment  
 365 is shown in blue.

366  
 367 *Configuration 2: Load at Midspan, Weight at Top:*

368 As before, solving Equation 11 for loading configuration 2 at the weight's location results  
 369 in:

$$370 \quad \begin{bmatrix} 1 & \frac{-h_1}{2EI} \\ -w & 1 \end{bmatrix} \begin{bmatrix} \delta_2 \\ W \end{bmatrix} = \begin{bmatrix} \frac{Fh_2^2(3h_1 - h_2)}{6EI} \\ 0 \end{bmatrix} \quad (19)$$

371 Solving for the weight-induced moment and solving for Equation 5 for the point of  
 372 loading ( $\delta_1$ ) to find a relationship between the test data and the flexural stiffness of the stem:

$$373 \quad \delta_2 = \frac{Fh_2^3}{3EI} + \frac{Fwh_2^4(3h_1 - h_2)}{6EI(2EI - wh_2^2)} \quad (20)$$

374 Where Equation 15 is shown in black, and the correction factor for the weight-induced moment  
 375 is shown in blue.

377 It should be noted that Equations 18 and 20 are simply Equation 15 with the addition of a  
378 correction factor that accounts for the influence of the weight-induced bending moment. Thus by  
379 comparing the results of Equation 15 with either Equation 18 or 20, the influence of the weight-  
380 induced bending moment on the displacement of the stem can be calculated. Additionally, the  
381 results of Equation 18 and 20 (i.e., the displacements) can be input into Equation 6 to determine  
382 the magnitude of the weight-induced moment. The weight induced bending moment ( $W$ ) can  
383 then be compared to the bending moment induced from the applied force ( $M_{ext}$ ) to determine the  
384 effect of self-weight on the total bending moment ( $M_{total}$ ) of the stem. Thus, researchers can  
385 easily determine if the weight-induced bending moments are negligible for their testing purposes,  
386 or if they need to be incorporated into the biomechanical models used in their studies.

387  
388 ***The Influence of Self-Loading for Various Plants***

389 As a point of reference, five plants – maize (*Zea mays*), wheat (*Triticum aestivum*), sweet  
390 sorghum (*Sorghum bicolor*), bamboo (*Bambusoideae*), and rice (*Oryza sativa*) – were analyzed  
391 to determine the impact of self-weight on plant bending moments and displacements. The values  
392 shown in Table 2 represent typical values reported in the literature for these plants. It should be  
393 noted that these are average single data points and a significant amount of variation in heights,  
394 weights, and flexural stiffnesses is expected within a given plant species. This information is  
395 presented here as an accessible reference for researchers to develop an understanding of the types  
396 of plants that are more or less affected by self-loading.

397  
398 ***Table 2: Self-loading related properties and the error of the induced moment at the base of the***  
399 ***stalk if self-loading is ignored. The center of gravity of the plant is assumed to be halfway up the***  
400 ***stem.***

| Plant         | Plant Height (mm) | Grain Height (mm) | Plant Weight (N) | Grain Weight (N) | Flexural Stiffness (Nm <sup>2</sup> ) | Error of M <sub>TOTAL</sub> at base (%) | Error of displacement at top (%) | References   |
|---------------|-------------------|-------------------|------------------|------------------|---------------------------------------|---|----------------------------------|--------------|
| Maize         | 2250              | 1125              | 7.595            | 2.874            | 79.17                                 | < 1%                                    | < 1%                             | [5,22,27–29] |
| Wheat         | 638               | 638               | 0.016            | 0.021            | 0.027                                 | 12%                                     | 16%                              | [23,30,31]   |
| Sweet Sorghum | 2650              | 2650              | 9.64             | 0.346            | 137.1                                 | 1%                                      | 1%                               | [32–34]      |
| Bamboo        | 10,774            | N/A               | 138.02           | N/A              | 229                                   | < 1%                                    | < 1%                             | [35,36]      |
| Rice          | 969               | 969               | 0.0635           | 0.028            | 0.17                                  | 6%                                      | 8%                               | [37,38]      |

401

402 A key factor in determining the influence of self-loading is the ratio of the weight of a  
403 plant to its flexural stiffness. Although this parameter does not include all of the factors that  
404 influence self-loading, it can be used as a quick evaluation tool for researchers to determine the  
405 general amount of influence self-loading may have. Figure 8 depicts the influence of this ratio  
406 on the induced moment and displacement, with the plant varieties in Table 2 shown as data  
407 points. In general, it can be seen that self-weight has a negligible effect on stiff and strong stems  
408 (i.e., bamboo and maize) but becomes more influential in smaller stems (i.e., rice, wheat). It  
409 should be noted that because flexural stiffness is defined as the externally applied load divided  
410 by the displacement, the displacement and flexural stiffness relate to each other with a -1:1 ratio,  
411 e.g. a 10% increase in the displacement for a given externally applied load results in a 10%  
412 decrease in the flexural stiffness. Thus, the error of displacement at the top of stem as shown in  
413 Table 2 can also be interpreted as the error in flexural stiffness.

414

#### 415 *The Influence of Self-Loading on Stalk Lodging Resistance*

416 The equations presented above quantify the influence of self-loading on the deflection  
417 and bending moment of plant stems. The association between deflections and bending moments

418 and stalk lodging resistance are plant- and time-specific. For instance, in late-season lodging of  
419 maize stalks, previous studies have found that the plant experience a predominantly linear-elastic  
420 response prior to failure, and that flexural stiffness tends to strongly correlate with lodging  
421 resistance [5]. In such a case, Equation 14 demonstrates that the total moment and bending stress  
422 are directly linear, e.g. a 10% increase in the total moment will result in a 10% increase in stress.  
423 Therefore, the authors hypothesize that increasing the induced bending moment will decrease the  
424 lodging resistance at a ratio of -1:1, e.g. a 10% increase in the induced bending moment from  
425 self-loading will result in a 10% decrease in the lodging resistance of the stalk. However, for  
426 less linear material responses (e.g. during green-snap), these relationships will be less direct. For  
427 stems with nonlinear material responses, researchers will need to incorporate these self-loading  
428 equations into their biomechanical models which contain non-linear material responses.

429         Based on prior experience and the results presented in Table 2 and Figure 8, the authors  
430 recommend that self-weight be accounted for when testing small grain stems. Several researchers  
431 have recognized that self-weight in small grains can introduce measurement errors. To minimize  
432 these errors researchers have normalized bending strength measurements by specimen weight.  
433 This was an important first step to properly accounting for self loading. However, in the future  
434 the authors recommend the equations and excel spreadsheet presented herein be used to properly  
435 account for the effect of self-weight. The effect of self-weight on the bending strength and  
436 flexural stiffness of small grain stems is complex and is not fully captured by simply normalized  
437 bending strength measurements by specimen weight. In general, the effect of self-weight on  
438 large grain stems such as maize is minimal and for many intents and purposes is most likely  
439 negligible.

440 ***Limitations***



441           The primary limitation of the current study is that the stalk was assumed to be in-line with  
442 the assumptions made for pure bending, including maintaining a constant cross-section with  
443 homogeneous, isotropic, linear elastic material subjected to pure bending [4]. The inclusion of the  
444 changes in cross-sectional geometry along the lengths of the stalk [22], material heterogeneity and  
445 anisotropy, or non-linear material properties would likely change the behavior of the analytical  
446 system. Further discussion of the influence of such material assumptions on equations has been  
447 investigated in a previous study by the authors [4]. These assumptions, when combined with the  
448 assumption of a single cross-section along the entire length of the stalk, results in a single flexural  
449 stiffness parameter for the entire stalk. However, the flexural stiffness of plants changes constantly  
450 along the length of the stalk (i.e., the diameter of most plant stems are large near the base of the  
451 plant and smaller near the top of the plant). The simplifying assumption of a single flexural  
452 stiffness parameter was deliberately made to allow for an easily-used generalized equation. This  
453 assumption is routinely made in phenotyping studies as well. If researchers need to incorporate  
454 changes in flexural stiffness along the length of the stalk, the approach presented in this study can  
455 be incorporated into a full Castigliano's method beam approximation [17]. Additionally, the  
456 equations used in this study assume small strains and small displacements. As such, these  
457 equations carry the same limitations as standard engineering beam bending equations, and are not  
458 suitable to predict post-failure loading conditions or displacements. When post-buckling analyses  
459 are required, non-linear finite element modeling approaches are recommended.

460           Finally, Equations 1, 2, and 6 assume that the maximum moment induced by self-loading  
461 is applied to the entire length of the stem below the weight, which is not accurate, and is used as a  
462 simple estimation of the moment induced by self-loading. In reality, self-loading is not a constant  
463 moment along the length of the stalk, but instead is an axial compressive load that induces a

464 moment that varies along the length of the stalk. However, modeling loading as an axial  
465 compressive load greatly increases the complexity of the equation, to the point that the matrix  
466 equations presented in this study would not be practical. Therefore, Equation 6 presents an upper-  
467 bound of the influence of self-loading by simply applying the maximum moment along the entire  
468 length of the stem. As shown in Figure 3 and Figure 4, this assumption is reasonable for the  
469 parameter space explored.

## 470 **Conclusions**

471 The self-loading of plants plays a potentially critical role in determining the flexural  
472 stiffness and bending strength of plant stems. To the best of the authors' knowledge, this is the  
473 first study that presents a methodology for (1) incorporating the self-loading of plants with  
474 externally applied forces, (2) investigating the interconnected nature of the loading conditions  
475 placed on the plant, and (3) exploring the interconnected parameters that impact the influence of  
476 self-loading on the plant's structural integrity. Two common phenotyping configurations were  
477 presented with simplified correction factors for researchers to use in their studies. A user-  
478 friendly Microsoft Excel framework was presented that allows researchers to quickly and easily  
479 determine the relevance of self-loading to their research experiments. A survey of five self-  
480 loaded plant species was presented to enable researchers to develop an intuition on the level of  
481 influence self-loading plays on the structural integrity of various plants. It is the  
482 recommendation of the authors that self-loading be taken into account for plants such as wheat  
483 and rice that have a large ratio of weight to flexural stiffness.

## 484 **Declarations**

485 **Ethics Approval and Consent to Participate**

486 Not applicable

487 **Consent for Publication**

488 Not applicable.

489 **Availability of Data and Materials**

490 The datasets used and/or analyzed during the current study are available from the  
491 corresponding author on reasonable request.

492 **Competing Interests**

493 The authors declare that they have no competing interests

494 **Funding**

495 This work was funded in part by the National Science Foundation (Award #1826715) by  
496 the United States Department of Agriculture - NIFA (#2016-67012-2381) and by the Canada First  
497 Research Excellence Fund (CFREF). Any opinions, findings, conclusions, or recommendations  
498 are those of the author(s) and do not necessarily reflect the view of the funding bodies.

499 **Authors' Contributions**

500 All authors were fully involved in the study and preparation of the manuscript. The  
501 material within has not been and will not be submitted for publication elsewhere.

502 **Acknowledgements**

503           Field data collection was completed by Undergraduate Research Assistants Matthew  
504 Kolbeck and Jonathan Fenske at the University of Saskatchewan's Plant Phenotyping and Imaging  
505 Research Centre (P2IRC).

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607

608 **Figure 1:** The loading diagram of a deflected stem, showing two loading locations with all three  
609 types of loading (an applied force, an applied moment, and a weight).

610

611 **Figure 2:** To obtain biomass samples a square sample area (a) was chosen and then cleared of  
612 surrounding plants. The sampling boundary height was measured (b) and then the biomass above  
613 this layer was cut and bagged. This was continued in 10 cm layers until there was no biomass left  
614 (c).

615

616 **Figure 3:** A comparison between the closed form solution and finite element models for the  
617 displacement at the top of the specimen (a) and the total moment at the base of the specimen (b),  
618  $n=768$ ; A histogram of the error between the closed form solution and the finite element models

619 for the displacement at the top of the specimen (c) and the total moment at the base of the specimen  
620 (d), n= 768.

621

622 **Figure 4:** A comparison between the closed form solution and finite element model for  
623 displacements beyond loading that would typically be seen in the field. Plots depict the  
624 displacement (normalized to the height of the stem) at the top of the specimen (a) and the total  
625 moment (normalized to the maximum moment) at the base of the specimen (b); the error in the  
626 closed form displacement and total moment values for large displacements, with the finite element  
627 model displacements shown (c).

628 **Figure 5:** An example of the Excel spreadsheet in the Supplementary Data, showing loading at  
629 three locations, and calculating displacement and induced moments at four locations: the three  
630 loading locations and the base of the plant. Note that error in displacement is not calculated at the  
631 base, as displacement at the base is zero regardless of loading condition.

632

633 **Figure 7:** The loading diagrams for two common mechanical phenotyping test protocols used to  
634 determine flexural stiffness; a typical maize phenotyping protocol (left), and a typical wheat  
635 phenotyping protocol (right).

636

637 **Figure 6:** The contribution of self-loading for the displacement at the top of the stem (left) and the  
638 total moment at the base of the stem (right), for both wet and dry stems.

639

640 **Figure 8:** The error of displacement at the top of the stem (left) and  $M_{\text{TOTAL}}$  at the base of the stem  
641 (right), as a function of the ratio between the combined weight of the grain and plant and the  
642 flexural stiffness of the stem.